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The Structure of Radiative Shocks

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- Non-trivial, analytic solutions do not exist for radiation hydrodynamics.
 - Compromise: Generate *semi-analytic* solutions; solutions from numerically integrating nonlinear ODEs.
 - Seek traveling wave solutions \Rightarrow *radiative shocks*.
- In this talk, use solutions to
 - verify code correctness
 - test AMR (rad-shocks are multiscale problems)
- Gives physics insight that would be *very* difficult to obtain by a series of runs from a computational physics code.



Outline



- 2 Equations of radiation hydrodynamics
 - Grey nonequilibrium diffusion
 - Euler coupled with grey nonequilibrium diffusion
 - Equilibrium diffusion limit
- 3 Overview of Semi-analytic Approach
- Sample Solutions
- 5 Code Comparison



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Overview of radiative shocks

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As one example, regimes in astrophysics governed by

- Inviscid hydrodynamics (Euler equations; non-relativistic)
- Thermal radiation (X-ray) transport
- High-energy density:
 - Material temperatures O(1 keV = 11.6 × 10⁶ Kelvin)
 - Radiation pressure affects hydrodynamics



Past Work on Radiative Shocks

 Most theory on "thick – thick" shocks: equilibrates on either side of shock.

Overview of Theory:

- Y. B. ZEL'DOVICH and Y. P. RAIZER, Physics of Shock Waves and High-Temperature Hydrodynamic Phenomena, 1966 (Dover 2002).
- D. MIHALAS and B. W. MIHALAS, Foundations of Radiation Hydrodynamics, 1984 (Dover 1999).
- R.P. DRAKE, High-Energy-Density Physics, Springer, 2006.

Grey VEF/AMR, ion/electron, calculations:

M. W. SINCELL, M. GEHMEYR, and D. MIHALAS, two articles in Shock Waves, 1999.

Theory and Solutions:

- Equilibrium: S. BOUQUET, R. TEYSSIER, and J. P. CHIEZE, Astrophysical Journal Supplement Series, 2000.
- Equilibrium: R. B. LOWRIE and R. M. RAUENZAHN, Shock Waves, 2007.
- Nonequilibrium: R. B. LOWRIE and J. D. EDWARDS, Shock Waves, 2008.

Extended thermodynamics:

W. Weiss, "Structure of Shock Waves," in Rational Extended Thermodynamics, Müller & Ruggeri (eds.), Springer, 1998.

Inviscid Hydrodynamic Shocks



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Alamo

Subcritical Radiative Shocks ($T_p < T_1$)



Supercritical Radiative Shocks ($T_p = T_1$)



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Zel'dovich Spike



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Grey nonequilibrium diffusion in 1 slide 1-D planar (slab) geometry

First two moments of radiative transfer equation:

$$\partial_t E + \partial_x F = c\sigma_a[B(T) - E] - \frac{\sigma_t}{c} \left(F - \frac{4}{3}vE\right)v$$
$$\frac{1}{c^2}\partial_t F + \partial_x P = -\frac{\sigma_t}{c} \left(F - \frac{4}{3}vE\right)$$

Nonequilibrium diffusion sets P = E/3 and drops $\partial_t F$ to yield

$$\partial_t E + \frac{4}{3} \partial_x (vE) - \partial_x \left(\frac{c}{3\sigma_t} \partial_x E \right) = c\sigma_a [B(T) - E] + \frac{1}{3} v \partial_x E$$

Underlined terms often dropped in "low-energy density" approximation.



Euler Coupled with Grey Nonequilibrium Diffusion

Nondimensional form, 1-D planar (slab) geometry

Most of this talk will be about solutions of

$$\begin{aligned} \partial_t \rho + \partial_x (\rho \mathbf{v}) &= \mathbf{0}, \\ \partial_t (\rho \mathbf{v}) + \partial_x \left(\rho \mathbf{v}^2 + \mathbf{p} + \frac{1}{3} \mathcal{P}_0 T_r^4 \right) &= \mathbf{0}, \\ \partial_t (\rho \mathcal{E}) + \partial_x \left[\mathbf{v} (\rho \mathcal{E} + \mathbf{p}) \right] &= \mathcal{P}_0 \sigma_a (T_r^4 - T^4) - \frac{1}{3} \mathcal{P}_0 \mathbf{v} \partial_x T_r^4, \\ \partial_t (\rho \mathcal{E} + \mathcal{P}_0 T_r^4) + \partial_x \left[\mathbf{v} \left(\rho \mathcal{E} + \mathbf{p} + \frac{4}{3} \mathcal{P}_0 T_r^4 \right) \right] &= \mathcal{P}_0 \partial_x (\kappa \partial_x T_r^4), \end{aligned}$$

where $\mathcal{E} = e + \frac{1}{2}v^2$ and for a γ -law EOS

$$p = \frac{\rho T}{\gamma}, \quad e = \frac{T}{\gamma(\gamma - 1)}, \quad \mathcal{P}_0 = \frac{\tilde{a}_R \tilde{T}_0^4}{\gamma \tilde{p}_0} \approx \frac{\text{rad. pressure}}{\text{mat. pressure}}.$$
Embedded discontinuities are hydrodynamic shocks.

Equilibrium Diffusion Limit

We'll also discuss "1T" solutions. Optically thick limit, $T_r \rightarrow T$, and our system reduces to

$$\partial_t \rho + \partial_x (\rho \mathbf{v}) = \mathbf{0},$$

$$\partial_t (\rho \mathbf{v}) + \partial_x \left(\rho \mathbf{v}^2 + \mathbf{p} + \frac{1}{3} \mathcal{P}_0 \mathbf{T}^4 \right) = \mathbf{0},$$

$$\partial_t (\rho \mathbf{E} + \mathcal{P}_0 \mathbf{T}^4) + \partial_x \left[\mathbf{v} \left(\rho \mathbf{E} + \mathbf{p} + \frac{4}{3} \mathcal{P}_0 \mathbf{T}^4 \right) \right] = \mathcal{P}_0 \partial_x (\kappa \partial_x \mathbf{T}^4).$$

- Euler equations with nonlinear heat conduction and a modified equation-of-state.
- For small- \mathcal{P}_0 , heat conduction may dominate.
- Embedded discontinuities are isothermal shocks.



Problem statement

- Assume that far from shock, $T_r = T$ ("thick–thick").
- Galilean invariant, so use steady frame.
- *Given:* The value γ and
 - Pre-shock state $(x \rightarrow -\infty)$: ρ_0 , T_0 , $T_{r0} = T_0$.
 - ▶ Non-dimensional constants: \mathcal{P}_0 and \mathcal{M}_0 ($\equiv \tilde{v}_0/\tilde{a}_0$, Mach number).
 - Functions $\sigma_a(\rho, T)$ and $\kappa(\rho, T)$ ($\tilde{\kappa} = \tilde{c}/3\tilde{\sigma}_t$).
- *Calculate:* $\rho(x)$, v(x), T(x), and $T_r(x)$.
- Optional: Transform back to a frame where shock is moving.





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Euler Coupled with Grey Nonequilibrium Diffusion

Nondimensional form, 1-D planar (slab) geometry

Seek traveling-wave solutions of

$$\partial_{t}\rho + \partial_{x}(\rho \mathbf{v}) = \mathbf{0},$$

$$\partial_{t}(\rho \mathbf{v}) + \partial_{x}\left(\rho \mathbf{v}^{2} + \mathbf{p} + \frac{1}{3}\mathcal{P}_{0}T_{r}^{4}\right) = \mathbf{0},$$

$$\partial_{t}(\rho E) + \partial_{x}\left[\mathbf{v}(\rho E + \mathbf{p})\right] = \mathcal{P}_{0}\sigma_{a}(T_{r}^{4} - T^{4}) - \frac{1}{3}\mathcal{P}_{0}\mathbf{v}\partial_{x}T_{r}^{4},$$

$$\partial_{t}(\rho E + \mathcal{P}_{0}T_{r}^{4}) + \partial_{x}\left[\mathbf{v}\left(\rho E + \mathbf{p} + \frac{4}{3}\mathcal{P}_{0}T_{r}^{4}\right)\right] = \mathcal{P}_{0}\partial_{x}(\kappa\partial_{x}T_{r}^{4}),$$



< 17 ▶

Overall Jump Relation

Integrate conservation equations from $-\infty < x < \infty$ to give:

$$\begin{pmatrix} \rho \mathbf{v} \\ \rho \mathbf{v}^2 + \mathbf{p}^* \\ (\rho \mathbf{E}^* + \mathbf{p}^*) \mathbf{v} \end{pmatrix}_0 = \begin{pmatrix} \rho \mathbf{v} \\ \rho \mathbf{v}^2 + \mathbf{p}^* \\ (\rho \mathbf{E}^* + \mathbf{p}^*) \mathbf{v} \end{pmatrix}_1,$$

where

$$m{
ho}^* = m{
ho} + rac{1}{3} \mathcal{P}_0 T^4, \quad m{e}^* = m{e} + rac{1}{
ho} \mathcal{P}_0 T^4, \quad E^* = m{e}^* + rac{1}{2} v^2.$$

- Get ninth-order polynomial in *T*₁; see Bouquet *et al* (2000).
- Same procedure used for any radiation model.

Hydro Shock Relations

At a discontinuity separating state- $_p$ and state- $_s$:

$$\begin{pmatrix} \rho \mathbf{v} \\ \rho \mathbf{v}^2 + \mathbf{p} \\ (\rho \mathbf{E} + \mathbf{p}) \mathbf{v} \\ -\kappa \partial_x T_r^4 + \frac{4}{3} \mathbf{v} T_r^4 \end{pmatrix}_{\mathbf{p}} = \begin{pmatrix} \rho \mathbf{v} \\ \rho \mathbf{v}^2 + \mathbf{p} \\ (\rho \mathbf{E} + \mathbf{p}) \mathbf{v} \\ -\kappa \partial_x T_r^4 + \frac{4}{3} \mathbf{v} T_r^4 \end{pmatrix}_{\mathbf{p}}^{\mathbf{a}}$$

- First 3 equations: Standard hydro jump conditions. Holds for any radiation model.
- Last equation: Continuity of *Eulerian frame* radiation flux.



Reduced Equations

4 PDEs reduce to 2 ODEs. We use Mach number (\mathcal{M}) as the independent variable:

$$\frac{dx}{d\mathcal{M}} = \frac{3\mathcal{M}_0(\mathcal{M}^2 - 1)\rho\beta}{\mathcal{P}_0},$$
$$\frac{dT}{d\mathcal{M}} = (\gamma - 1)\beta \left[4\mathcal{M}_0 T_r^3 T_r' + (\gamma \mathcal{M}^2 - 1)r\right],$$

where $\beta(T, M)$ and r(T, M) are known functions, and

$$\rho(T,\mathcal{M}) = \frac{\mathcal{M}_0}{\mathcal{M}\sqrt{T}}, \quad T_r(T,\mathcal{M}) = \frac{1}{\gamma \mathcal{P}_0} \left[\mathcal{K}_m - 3\gamma \frac{\mathcal{M}_0^2}{\rho(T,\mathcal{M})} - 3T\rho(T,\mathcal{M}) \right]$$

• If $T_r \approx \text{const.}$, then $\mathcal{M} = 1/\sqrt{\gamma}$ corresponds to maximum T (isothermal sonic point – ISP).

Overview of Solution Procedure

- Compute post-shock state $(x \to \infty)$: Find root of a ninth-order polynomial in T_1 .
- Ind states are typically saddle points. Solve two IVPs:
 - Find precursor region: Integrate ODEs from $\mathcal{M} = \mathcal{M}_0$ to $\mathcal{M} = 1$.
 - Find relaxation region: Integrate ODEs from $\mathcal{M} = \mathcal{M}_1$ to $\mathcal{M} = 1$.
- If the two ODE solutions do not match at M = 1, then shift the solutions such that they are connected with a hydro shock.

See Lowrie & Edwards (Shock Waves, 2008) for the details.





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Radiative shock solutions may be characterized by the following:

- Is there an embedded hydro shock?
 - If no, all variables continuous.
 - If yes,
 - ★ Subcritical ($T_p < T_1$).
 - * Supercritical ($T_{\rho} \approx T_1$).
- Is there an isothermal sonic point (ISP; $M = 1/\sqrt{\gamma}$)?
 - If yes, then there is a temperature (Zel'dovich) spike $\implies T_{max} > T_1$.
 - Lowrie & Edwards (2008) derive an algebraic expression for T_{max} that is very accurate.
- May have a hydro shock without a spike.
- May have a spike without a hydro shock.

A (10) A (10)

Isothermal Shock (Equilib.) ⇔ Spike (Non-equilib.)

 $\gamma = 5/3$. See Lowrie & Rauenzahn (2007).



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 $\mathcal{M}_0 = 1.05$ solution $\mathcal{P}_0 = 10^{-4}, \sigma_a = 10^6, \kappa = 1, \gamma = 5/3$; subcritical; no embedded shock or isothermal sonic point (ISP).



$\mathcal{M}_0 = 1.2 \text{ solution}$

Subcritical; shock, but no ISP.



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$\mathcal{M}_0 = 2$ solution

Subcritical; ISP coincident with shock.



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$\mathcal{M}_0 = 3$ solution

Subcritical; ISP downstream of shock.



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$\mathcal{M}_0=3 \text{ solution}$

Spike region.



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$\mathcal{M}_0=5 \text{ solution}$

Supercritical; ISP downstream of shock.



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$\mathcal{M}_0 = 5$ solution (continued)

Zel'dovich spike region



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Isothermal Shock (Equilib.) \Leftrightarrow Spike (Non-equilib.)

 $\gamma = 5/3$. See Lowrie & Rauenzahn (2007).



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$\mathcal{M}_0 = 27$ solution

Supercritical; ISP downstream of shock. NOTE OVER COMPRESSION.



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$\mathcal{M}_0=27$ solution (continued)

Zel'dovich spike region



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$\mathcal{M}_0=30 \text{ solution}$

No shock, but still an ISP!



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$\mathcal{M}_0 = 30 \text{ solution (continued)}$ Spike region



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$\mathcal{M}_0 = 50 \text{ solution}$ No shock or ISP



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Code Comparison Test Cases

- Hydrogen gas, $\gamma = 5/3$
- Bremsstrahlung absorption model (Zel'dovich & Razier; assume fully ionized):

$$\sigma_a(\rho, T) = \sigma_{a,0}\rho^2 T^{-7/2}$$

• Thomson scattering:

$$\sigma_{s}(\rho) = \sigma_{s,0}\rho$$

- $ho_0 = 1 \text{ g/cc}$
- Cases:

1
$$T_0 = 10 \text{ eV}, \mathcal{M}_0 = 10, \text{ subcritical}$$

- 2 $T_0 = 100 \text{ eV}, \mathcal{M}_0 = 5$, supercritical
- **(3)** $T_0 = 100 \text{ eV}, M_0 = 45$, no embedded hydro shock
- Comparison of relaxation rates shows that 3T effects should be small.
- Compare with a finite-volume, Godunov-based AMR code (RAGE).



Two Initialization Methods

1

Initialize with exact shock profile.

- How well can the code propagate the shock and maintain the profile?
- Pros: Boundary conditions not an issue.
- Cons: Requires code initialize from exact solution.
- This method was used for all error-norm calculations.
- Piston problem.
 - Black-body piston moving at v_1 , radiating at $T_r = T_1$.
 - Given enough time, the resulting shock should match the semi-analytic solution.
 - Pros: Easy setup; better test.
 - Cons: Requires "large" solution domain; boundary conditions become an issue.

With either method, you should propagate the shock long enough to establish a self-similar profile.

 $T_0 = 10 \text{ eV}, M_0 = 10 \text{ Sample Results}$ $\Delta x = 50 \ \mu\text{m}, \text{ propagated 1 cm}, T_1 \approx 321 \text{ eV}, T_{\text{max}} \approx 419 \text{ eV}$



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$T_0 = 10 \text{ eV}, M_0 = 10 \text{ Error Convergence}$

Exact solution computed without flux limiter.



 $T_0 = 10 \text{ eV}, M_0 = 10, \text{Piston Problem Initialization}$ Legend indicates distance exact profile has moved. Solutions shifted to align. $\Delta x = 50 \ \mu \text{ m}.$



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Profile vs. Piston Initialization



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$T_0 = 100 \text{ eV}, M_0 = 5 \text{ Sample Results}$

AMR (8 levels): $\Delta x_{max} = 5$ cm, $\Delta x_{min} \approx 391 \ \mu$ m, propagated 230 cm, $T_1 \approx 857$ eV, $T_{max} \approx 1.08$ keV



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$T_0 = 100 \text{ eV}, \mathcal{M}_0 = 5 \text{ Spike Region}$



• LOS Alamos

$T_0 = 100 \text{ eV}, M_0 = 5 \text{ Refinement Distribution}$

8 levels \implies 814 cells; 10 levels \implies 528 cells (10-level mesh derefined in precursor).



• LOS Alamos

Refinement and Error Distribution

Even though 10-level mesh derefined in $27 \leq x \leq 49$, it is more accurate.





 $T_0 = 100 \text{ eV}, \mathcal{M}_0 = 45 \text{ Sample Results}$ $\Delta x = 1.25 \text{ cm}, \text{ propagated } 20 \text{ m}, T_1 \approx 8.36 \text{ keV}, T_{\text{max}} \approx 8.48 \text{ keV}$



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 $T_0 = 100 \text{ eV}, M_0 = 45 \text{ Spike Region}$ RAGE results with $\Delta x = 1.25 \text{ cm}/2^N$, with N = 0, 1, 2, 3



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$T_0 = 100 \text{ eV}, \mathcal{M}_0 = 45 \text{ Convergence}$

A hard-coded tolerance was limiting the convergence rate.



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• The radiative shock solutions test several issues:

- Fully-coupled radiation hydrodynamics
- Smooth problems \implies ability to attain theoretical convergence rate
- Refinement criteria for AMR (solutions are multiscale)
- Other radiation models
- Future work
 - Separate ion/electron temperatures
 - More advanced radiation models

